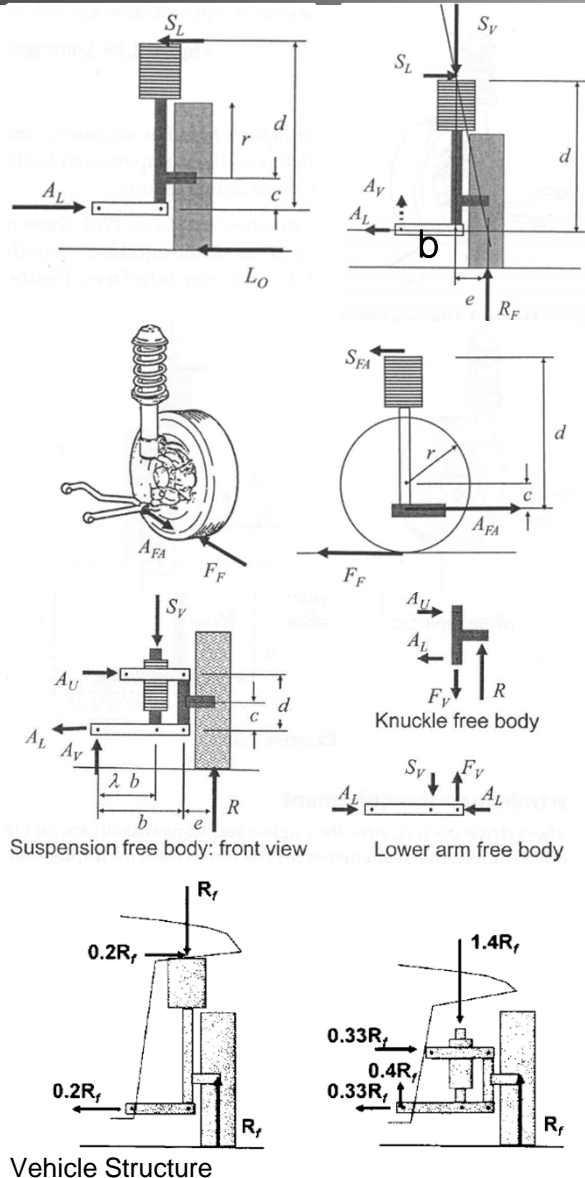


2.5 Struct vs.SLA



Struct

For the maximum lateral tire patch load during rollover mode

$$M_{@A_L} : S_L d - (r-c)L_0 = 0$$

$$\rightarrow S_L = \left(\frac{r-c}{d}\right)L_0$$

$$M_{@S_L} : A_L d - (d+(r-c))L_0 = 0$$

$$\rightarrow A_L = \left(\frac{r-c}{d} + 1\right)L_0$$

For the maximum fore-aft tire patch load during braking mode

$$M_{@A_{FA}} : F_F (r-c) - dS_{FA} = 0$$

$$\rightarrow S_{FA} = \left(\frac{r-c}{d}\right)F_F$$

$$M_{@S_{FA}} : A_{FA} d - (d+(r-c))F_F = 0$$

$$\rightarrow A_{FA} = \left(\frac{r-c}{d} + 1\right)F_F$$

For the maximum vertical tire patch load during bump mode

$$F_x : A_L - S_L = 0 \rightarrow S_L = A_L$$

$$F_y : R_F - S_V - A_V = 0 \rightarrow S_V = R_F - A_V$$

if $A_V = 0$ (or $S_V \gg A_V$), $S_V = R_F$

SLA

$$1) S_V$$

$$F_x : A_U - A_L = 0 \rightarrow A_U = A_L$$

$$F_y : S_V - A_V - R = 0 \rightarrow S_V = R + A_V$$

$$M_{@lowerarm, F_V} : bA_V - (b-\lambda b)S_V = 0 \rightarrow A_V = (1-\lambda)S_V$$

$$\rightarrow S_V - R = (1-\lambda)S_V \rightarrow S_V = \frac{1}{\lambda}R$$

$$2) A_U$$

$$M_{@knuckle, F_V} : eR - dA_U = 0 \rightarrow A_U = \frac{e}{d}R$$

$$3) A_L, A_V$$

$$A_L = A_U = \frac{e}{d}R$$

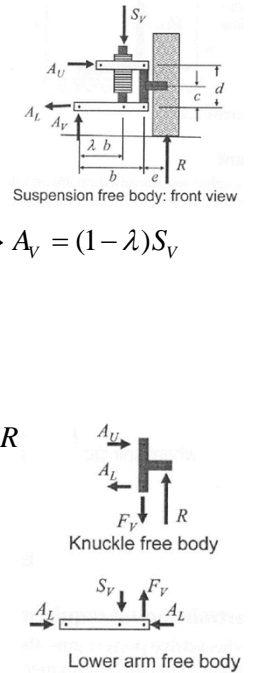
$$A_V = A_V = (1-\lambda)S_V = (1-\lambda)\frac{1}{\lambda}R$$

$$= \left(\frac{1}{\lambda} - 1\right)R$$

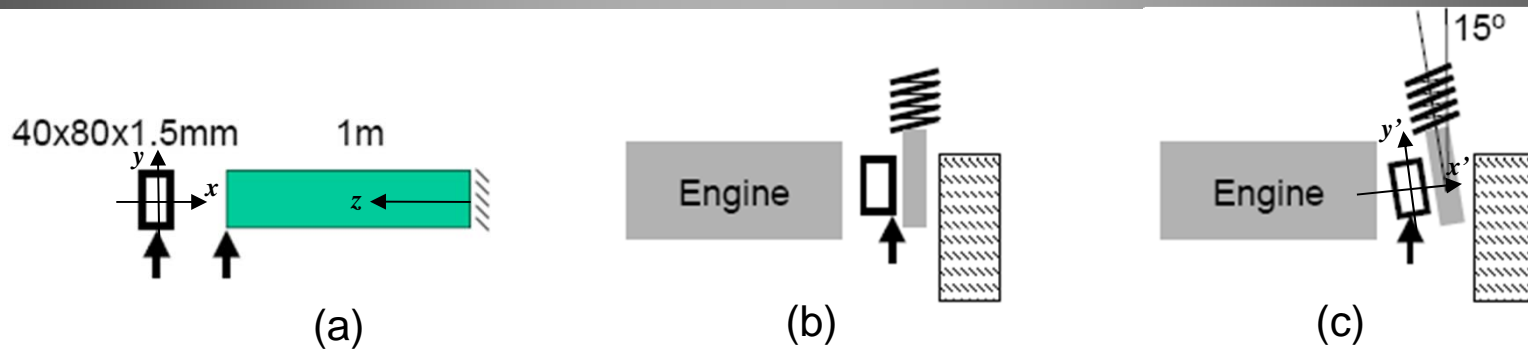
$\lambda = 0.7$ 인 경우

$$\text{Strut} : S_V = R, \text{ SLA} : \frac{1}{\lambda}R = \frac{1}{0.7}R \approx 1.43R$$

SLA가 Struct보다 Sv (spring-shock maximum load)가 1.43배 커서 우수함



3.2 Motor Compartment Rail (1)



(a) load at the centroid

$$I_x = \frac{40(80)^3}{12} - \frac{37(77)^3}{12} = 2.9902 \times 10^5 \text{ mm}^4$$

$$\sigma_z = -\frac{My}{I_x} = -\frac{(2000\text{N})(1000\text{mm})(40\text{mm})}{2.9902 \times 10^5 \text{ mm}^4} = -267.5 \text{ N/mm}^2$$

$$\delta = \frac{Fl^3}{3EI} = \frac{(2000\text{N})(1000\text{mm})^3}{3(207 \times 10^3 \text{ N/mm}^2)(2.9902 \times 10^5 \text{ mm}^4)} = 10.4 \text{ mm}$$

(b) load at the corner → load at the centroid + torsion(clockwise)

$$I_x = \frac{40(80)^3}{12} - \frac{37(77)^3}{12} = 2.9902 \times 10^5 \text{ mm}^4$$

$$\sigma_z = -\frac{My}{I_x} = -\frac{(2000\text{N} \times 1000\text{mm})(40\text{mm})}{2.9902 \times 10^5 \text{ mm}^4} = -267.5 \text{ N/mm}^2$$

$$\delta_y = \frac{Fl^3}{3EI} = \frac{(2000\text{N})(1000\text{mm})^3}{3(207 \times 10^3 \text{ N/mm}^2)(2.9902 \times 10^5 \text{ mm}^4)} = 10.8 \text{ mm}$$

$$\tau_{xy} = \frac{T}{2At} = \frac{(2000\text{N} \times 20\text{mm})}{2(38.5 \times 78.5 \text{ mm}^2)(1.5\text{mm})} = 4.4 \text{ N/mm}^2$$

$$\theta = \frac{Tl}{GJ_{eff}} = \frac{Tl}{G \frac{4A^2t}{S}} = \frac{(2000\text{N} \times 20\text{mm})(1000\text{mm})2(38.5 + 78.5\text{mm})}{4(80 \times 10^3 \text{ N/mm}^2)(38.5 \times 78.5 \text{ mm}^2)^2(1.5\text{mm})} = 0.002 \text{ rad}$$

$$\delta_{total} = \delta_y + \sqrt{20^2 + 40^2} \theta$$

3.2 Motor Compartment Rail (2)

(c) section rotated 15°

$$I_{x'} = \frac{40(80)^3}{12} - \frac{37(77)^3}{12} = 2.9902 \times 10^5 \text{ mm}^4$$

$$I_{y'} = \frac{80(40)^3}{12} - \frac{77(37)^3}{12} = 1.0164 \times 10^5 \text{ mm}^4$$

$$F_{y'} = F \cos 15^\circ = 1931.9$$

$$F_{x'} = F \sin 15^\circ = 517.6$$

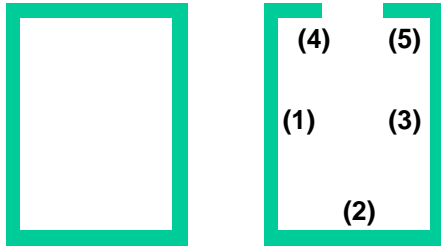
$$\sigma_z = -\frac{M_{y'} y'}{I_{x'}} - \frac{M_{x'} x'}{I_{y'}} = -\frac{(1931.9 \text{ N} \times 1000 \text{ mm})(40 \text{ mm})}{2.9902 \times 10^5 \text{ mm}^4} - \frac{(517.6 \text{ N} \times 1000 \text{ mm})(20 \text{ mm})}{1.0164 \times 10^5 \text{ mm}^4} = -360.3 \text{ N/mm}^2$$

$$\delta_{y'} = \frac{F_{y'} l^3}{3EI_{y'}} = \frac{(1931.9 \text{ N})(1000 \text{ mm})^3}{3(207 \times 10^3 \text{ N/mm}^2)(2.9902 \times 10^5 \text{ mm}^4)} = 10.4 \text{ mm}$$

$$\delta_{x'} = \frac{F_{x'} l^3}{3EI_{x'}} = \frac{(517.6 \text{ N})(1000 \text{ mm})^3}{3(207 \times 10^3 \text{ N/mm}^2)(1.0164 \times 10^5 \text{ mm}^4)} = 8.2 \text{ mm}$$

$$\delta = \sqrt{(\delta_{x'})^2 + (\delta_{y'})^2} = 13.2 \text{ mm}$$

3.16 Effective Width



$$\sigma_{cr} = k \frac{E\pi^2}{12(1-\nu^2)} \frac{1}{(b/t)^2} = 4 \frac{\pi^2 (207 \times 10^3 \text{ N/mm}^2)}{12(1-0.3^2)(100\text{mm}/0.86\text{mm})^2} = 53.35 \text{ N/mm}^2$$

$$I = \frac{100(150)^3}{12} - \frac{98.28(148.28)^3}{12} = 1.4237 \times 10^6 \text{ mm}^4$$

$$\sigma_{cr} = \frac{M_{cr}(75\text{mm})}{I} \rightarrow M_{cr} = 1.0128 \times 10^6 \text{ Nmm}$$

$$w = \frac{1}{2} \left(1 + \frac{\sigma_{cr}}{\sigma_s} \right) b \rightarrow \begin{cases} \sigma_s = 1.1\sigma_{cr} : w = 95.5\text{mm} \\ \sigma_s = 1.5\sigma_{cr} : w = 83.3\text{mm} \\ \sigma_s = 2.0\sigma_{cr} : w = 75.0\text{mm} \end{cases}$$

(lx)nom
1.42E+06

b	100	mm
h	150	mm
t	0.86	mm
w	75	mm

part	width	height	x	y	area	x*area	y*area	lx'	d	lx
1	0.86	150.00	0.43	75.00	129.00	55.47	9675.00	2.42E+05	3.95	2.44E+05
2	98.28	0.86	50.00	0.43	84.52	4226.04	36.34	5.21E+00	-70.62	4.21E+05
3	0.86	150.00	99.57	75.00	129.00	12844.53	9675.00	2.42E+05	3.95	2.44E+05
4	36.64	0.86	19.18	149.57	31.51	604.37	4713.01	1.94E+00	78.52	1.94E+05
5	36.64	0.86	80.82	149.57	31.51	2546.67	4713.01	1.94E+00	78.52	1.94E+05
sum					405.54	20277.08	28812.37			1.30E+06
centroid			50.00	71.05				lx/(lx)nom		0.91

(σ)y	207	N/mm ²
y	78.95	mm
My	3.40E+06	Mmm

4.2 Body Bending: Backbone Structure

$$(a) \text{ strength requirement: } \sigma_{allow} = \frac{Mc}{I} = \frac{\left(\frac{F_{max} L}{4}\right)c}{I} \rightarrow I_{strength} = \frac{LcF_{max}}{4\sigma_{allow}} = \frac{(2790mm)(150mm)(6670N)}{4(175N/mm^2)} = 3.99 \times 10^6 \text{ mm}^4$$

$$I_{strength} = \frac{h^2 t (3w + h)}{6} \rightarrow t = \frac{6I_{strength}}{h^2 (3w + h)} = \frac{6(3.99 \times 10^6 \text{ mm}^4)}{(300mm)^2 (3 \times 200mm + 300mm)} = 0.295mm$$

$$(b) \text{ stiffness requirement: } \delta = \frac{F_{max} L^3}{48EI} \rightarrow I_{stiffness} = \frac{F_{max} L^3}{48E\delta} = \frac{(6670N)(2790mm)^3}{48(207,000N/mm^2)(1mm)} = 1.46 \times 10^7 \text{ mm}^4$$

$$I_{stiffness} = \frac{h^2 t (3w + h)}{6} \rightarrow t = \frac{6I_{stiffness}}{h^2 (3w + h)} = \frac{6(1.46 \times 10^7 \text{ mm}^4)}{(300mm)^2 (3 \times 200mm + 300mm)} = 1.08mm$$

$$(f) I_{strength} = I_{stiffness} \rightarrow \frac{LcF_{max}}{4\sigma_{allow}} = \frac{F_{max} L^3}{48E\delta}$$

$$\rightarrow L = \sqrt{\frac{12E\delta c}{\sigma_{allow}}} = \sqrt{\frac{12(207,000N/mm^2)(1mm)(150mm)}{175N/mm^2}} = 1459mm (< 2790mm)$$

→ 2-seat car has balanced strength and stiffness requirements.

5.3 Body Torsion: Backbone Structure

$$(a) \text{ strength: } \tau_{allow} = \frac{T}{2At} \rightarrow t = \frac{T}{2A\tau_{allow}} = \frac{6,780,000 \text{ Nmm}}{2(200\text{mm} \times 300\text{mm})(86 \text{ N/mm}^2)} = 0.657 \text{ mm}$$

$$(b) \text{ stiffness: } \theta = \frac{Tl}{GJ_{eff}} \rightarrow k = \frac{T}{\theta} = \frac{GJ_{eff}}{l} \quad J_{eff} = \oint \frac{4A^2}{t} dS = \frac{4A^2 t}{S} \rightarrow k = \frac{T}{\theta} = \frac{G(4A^2 t)}{lS}$$

$$\rightarrow t = \frac{k l S}{4A^2 G} = \frac{\left(12,000,000 \frac{\text{Nmm}}{\text{deg}} \frac{360 \text{ deg}}{2\pi \text{ rad}}\right) (2790\text{mm}) 2(200\text{mm} + 300\text{mm})}{4(200\text{mm} \times 300\text{mm})^2 (83,000 \text{ N/mm}^2)} = 1.605 \text{ mm}$$

$$(c) t_{strength} < t_{stiffness}$$

→ The stiffness requirement is dominant assuming that the walls are stable and do not undergo plate buckling.

5.13 Torsion of Sedan

$$\begin{bmatrix} h_1 & w & 0 & 0 & 0 & 0 & 0 & 0 \\ l_S & 0 & -w & 0 & 0 & 0 & 0 & 0 \\ l_R & 0 & 0 & -w & 0 & 0 & 0 & 0 \\ l_B & 0 & 0 & 0 & -w & 0 & 0 & 0 \\ h_2 & 0 & 0 & 0 & 0 & w & 0 & 0 \\ l_F & 0 & 0 & 0 & 0 & 0 & -w & 0 \\ 0 & 0 & 0 & (h_0 - h_1) & [l_F(h_0 - h_2) + l_2(h_2 - h_1)]/l_B & -l_F & h_1 & 0 \end{bmatrix} \begin{bmatrix} Q_1 \\ Q_2 \\ Q_3 \\ Q_4 \\ Q_5 \\ Q_6 \\ Q_7 \end{bmatrix} = \begin{bmatrix} -T \\ 0 \\ 0 \\ 0 \\ -T \\ 0 \\ 0 \end{bmatrix} \begin{matrix} \text{dash} \\ \text{windshield} \\ \text{roof} \\ \text{backlight} \\ \text{rear seat panel} \\ \text{floor} \\ \text{side frame} \end{matrix}$$

$$\rightarrow \begin{bmatrix} 750 & 1500 & 0 & 0 & 0 & 0 & 0 & 0 \\ 707 & 0 & -1500 & 0 & 0 & 0 & 0 & 0 \\ 1250 & 0 & 0 & -1500 & 0 & 0 & 0 & 0 \\ 559 & 0 & 0 & 0 & -1500 & 0 & 0 & 0 \\ 750 & 0 & 0 & 0 & 0 & 1500 & 0 & 0 \\ 2000 & 0 & 0 & 0 & 0 & 0 & -1500 & 0 \\ 0 & 0 & 0 & 500 & 1789 & -2000 & 750 & 0 \end{bmatrix} \begin{bmatrix} Q_1 \\ Q_2 \\ Q_3 \\ Q_4 \\ Q_5 \\ Q_6 \\ Q_7 \end{bmatrix} = \begin{bmatrix} -T \\ 0 \\ 0 \\ 0 \\ -T \\ 0 \\ 0 \end{bmatrix}$$