

OPTISTRUCT FOR LINEAR ANALYSIS, V2019 CHAPTER 3: INERTIA RELIEF ANALYSIS



AGENDA

- 1. Introduction to Linear Analysis
 - Type of Analysis
 - Type of Elements and Materials
 - Type of Loads & Boundary Conditions
- 2. Linear Static Analysis
- 3. Inertia Relief Analysis
- 4. Modal Analysis
- 5. Linear Buckling Analysis
- 6. Thermal Stress Steady State Analysis

- 7. Advanced Topics
 - Debugging Guide
 - Parameters
 - Transitioning Elements
 - Introduction to Parallelization
 - Run Options
 - Output Management
- 8. Optimization in Linear Analysis
 - OptiStruct Optimization
 - DRCO Approach
 - Setting up Optimization
 - Optimization Responses for Linear Analysis

Inertia Relief analysis allows the simulation of unconstrained or partially-constrained structures.

Typical applications include:

- An airplane in flight
- Suspension parts of a car
- Satellite in space

This analysis type will calculate the steadystate stress and deformed shape in the structure as if it were freely accelerating due to the applied loads.



In OptiStruct, inertia relief can be applied to linear static and nonlinear gap analyses.

Inertia relief boundary conditions may be defined in the bulk data section of the input deck or they may be determined automatically by the solver:

Automatic Support generation with PARAM, INREL, -2

SUBCASE 6	<pre>\$ Defines subcase number 6</pre>
LOAD = 5	$\$ references all static load cards with ID = 5
BEGIN BULK	
PARAM, INREL, -2	\$ Defines inertia relief without support points
FORCE, 1, <mark>5</mark> ,	to the provide the set of the

Using SUPORT entries with PARAM, INREL, -1

- The SUPORT and SUPORT1 bulk data entries are used to define up to six reaction degrees of freedom of the free body.
- SUPORT entries will be used in all relevant subcases and therefore do not need to be referenced in the Subcase Information section.

\$ Defines subcase number 6
$\$ references all static load cards with SID = 5
\$ Defines inertia relief with a support point
\$ Defines force ID = 1 for load card ID = 5,
\$ Grid ID, DOF's supported

Using SUPORT entries with PARAM, INREL, -1

- The SUPORT and SUPORT1 bulk data entries are used to define up to six reaction degrees of freedom of the free body.
- SUPORT1 entries need to be referenced by a SUPORT1 data selector statement for use within a subcase.

SUBCASE 6	\$ Defines subcase number 6
LOAD = 5	$\$ references all static load cards with ID = 5
SUPORT1 = 8	\$ references all SUPORT1 cards with ID = 8
BEGIN BULK	
PARAM, INREL, -1	\$ Defines inertia relief with a support point
FORCE, 1, <mark>5</mark> ,	to the provide the set of the
SUPORT1,8,10,3	\$ SUPORT1 set = 8, Grid ID, DOF's Supported

HOW TO SETUP AN INERTIA RELIEF ANALYSIS

Inertia relief analyses can be defined in seven steps:

- Step 1 Generate a FEM model with materials & properties
- SUPORT or SUPORT1 cards (optional)
- Step 3 Define the equivalent static force to apply to the model
- Step 4 Define the INREL parameter to enable inertia relief calculation
- Step 5 Create the static load case including the fictitious supports
- Step 6 Run the analysis and post-process



File Name and Location

...\STUDENT-EXERCISE\3a_Satellite\satellite.hm

Objectives (1/2)

This exercise runs an inertia relief load case on a simple satellite. This is a test made with aerospace structures that will need to support inertia loads. The objective of this kind of test is to verify if the structure is strong enough to support these loads without a static failure.

- 1. Open the model in HyperMesh Desktop
- 2. Review the model and check total mass
- 3. Set common OUTPUT requests
- 4. Set control card PARAM, INREL to -1
- 5. Create a load collector Support (no card image) with the following SUPORT1 load type constraints representing the fictitious support
 - Node 2: DOF 2
 - Node 3: DOFs 1-3
 - Node 4: DOFs 1-2



Objectives (2/2)

- 6. Create the following load collectors (no card image) and forces on Node 1 of the model
 - 3x Gx: force components {92700,0,0} (equivalent to three times gravity in x direction)
 - 3x Gy: force components {0,92700,0} (equivalent to three times gravity in y direction)
 - 2x Gz: force components {0,0,61800} (equivalent to two times gravity in z direction)
- 7. Create a following load collector 4.7x G with card image LOADADD and select all three above created load collectors with scale factor 1.0
- 8. Create four Linear Static load cases using Support as the SUPORT1 entry for each of the four load collectors created before
- 9. Run the analysis with OptiStruct
- 10. Review the .out file wrt warnings and errors
- 11. Review the results in HyperView and check if
 - Max. relative displacement < 500 mm
 - Max. von Mises stress < 70 MPa for 2D modeled components



Hints (1/3)

2. Total mass: 3.090 t

area =		1		5	9	0	е	+	0	8
volume =		2		8	3	6	е	+	0	9
total mass =						3		0	9	0
	_	_	_	_	-					_

3. Use HyperMesh's Quick Access Tool (Crtl+f) to add control cards SCREEN OUT OUTPUT, H3D, ALL OUTPUT, HTML,, NO

4. Use HyperMesh's Quick Access Tool (Crtl+f) to set control card PARAM, INREL to -1

out			0,
GLOBAL_OUTPUT_REQUEST			
OUTFILE			
OUTPUT			
Setup Card Edit Output Blocks			
Setup Create Output Blocks			
Setup Delete Output Blocks			
Setup Edit Output Blocks			
Setup Organize Output Blocks			
Setup Renumber Output Blocks			
Entities	ID 💿	Include	
Eritaes		melade	
Beam Section Collectors (1)			
- 🐻 Cards (3)			
CREEN	1	0	
OUTPUT	2	0	
PARAM	3	0	
🕀 🛜 Components (14)			
🕀 🙀 Materials (4)			
🗄 😓 Properties (13)			
⊕ 📁 Titles (1)			
			-82
Name	Va	lue	- 10
I64SLV			
			- 10
INREL_VI		1	- 10
INTRFACE		ا م	

Name	Value
Include File	[Master Model]
Status	
number_of_outputs =	2
OUTPUT 1	
KEYWORD	H3D
FREQ	ALL
OPTION	
🗆 OUTPUT 2	
KEYWORD	HTML
OPTION	NO



Hints (2/3)

5. Switch the Load Type for constraint in HyperMesh from SPC to SUPORT1 before creating the constraints

6.-8. See screenshots



Name	Value
Solver Keyword	LOADADD
Name	4.7x G
ID	5
Color	
Include	[Master Model]
Card Image	LOADADD
User Comments	Hide In Menu/Export
S	1.0
LOAD_Num_Set =	3 LOAD_Num_Set
Data: S1,	
	51 LI
	1 1.0 (2) 3x Gx
	2 1.0 (3) 3x Gy
	3 1.0 (4) 2x Gz



— 📥 🛛 3x G x	1	0		
— 👍 🛛 Зх Бу	2	0		
	3	0		
👍 4.7x G	4	0		
·····				
me		Value		
Solver Keyword		SUBCASE		
Name		4.7x G		
ID		4		
Include File		[Master Model]		
User Comments		Hide In Menu/Export		
Subcase Definition				
😑 Analysis type		Linear Static		
SPC		<unspecified></unspecified>		
LOAD		4.7x G (5)		
SUPORT1		Support (1)		
PRETENSION		/Unspecified\		

Hints (3/3)

11. Max. relative displacement = 421.4 mm (load case 4.7x G) < 500 mm Max. von Mises stress = 22.3 MPa (load case 4.7x G) < 70 MPa



QUESTIONS & ANSWERS

- 1. Which of these describes the difference between linear static and inertia relief analysis?
 - a) Linear statics determines the vibrational response of a system, inertia relief uses static loads.
 - b) Inertia relief correctly solves unconstrained or partially-constrained structures, linear statics can not.
 - c) Linear statics solves unconstrained structures, linear statics is for partiallyconstrained structures

