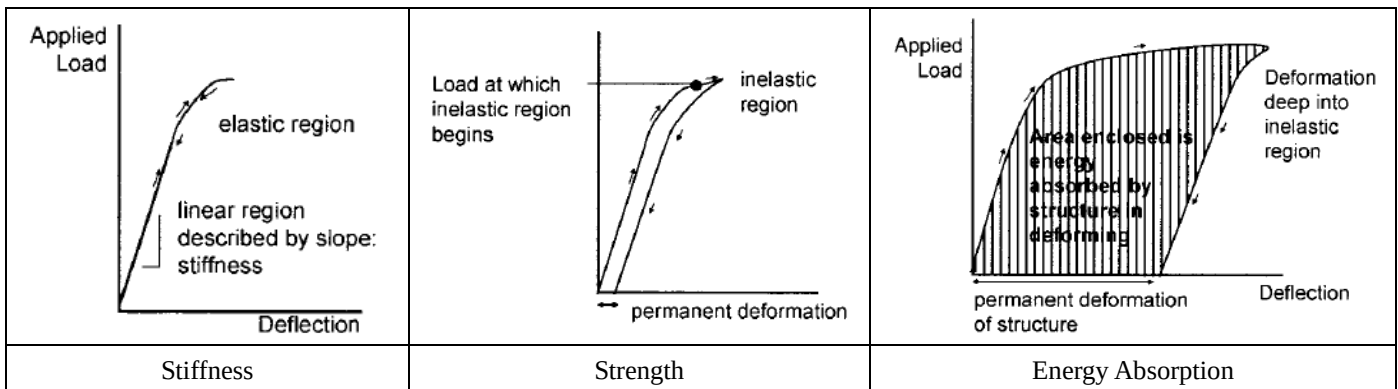


1. (2 pts each)

- | | |
|-----------------|-----------------------|
| (1) rocker | (2) midrail |
| (3) Shock tower | (4) Front roof header |
| (5) Roof rail | (6) Hinge pillar |
| (7) B pillar | (8) shear |
| (9) peel | (10) square |

2. (4 pts each)

- (1) Stiffness: Characterized by small and elastic deformation, (applied load)/(unit of deformation)
- (2) Strength: Characterized by the onset of a small permanent deformation, Lowest load at which a permanent deformation first appears
- (3) Energy Absorption: Characterized by very large permanent deformations where the amount of energy absorbed by the structure

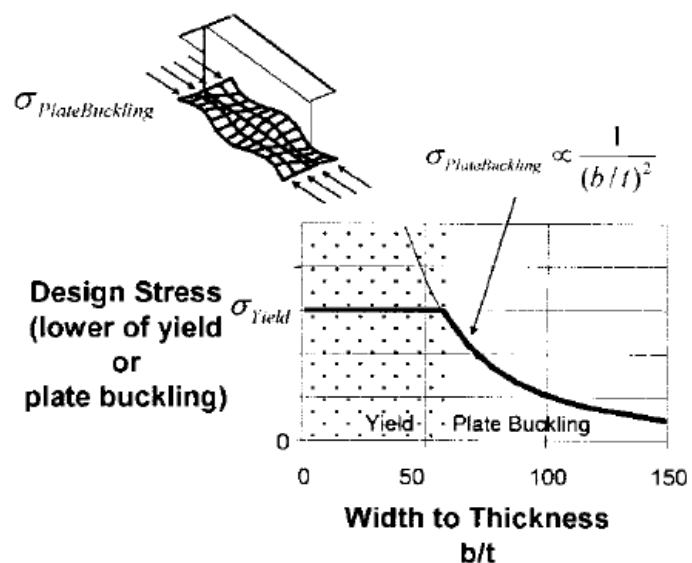


3. (4 pts each)

$$(1) \sigma_{CR} = \frac{E\pi^2}{12(1-\mu^2)(b/t)^2} k = \frac{748355}{(b/t)^2}$$

$$(2) \sigma_{design} = \begin{cases} \sigma_{yield} = 207 \text{ N/mm} \\ \sigma_{buckle} = \frac{748355}{(b/t)^2} \text{ N/mm} \end{cases}$$

$$\sigma_{yield} = \sigma_{buckle} \rightarrow (b/t) \approx 60$$



4.

(1) $X_d(2F)$, $X_{s1}(2F)$, $X_a(4/5F)$, $X_{s3}(8/5F)$, $X_b(8/5F)$, 각 1점

System free body $\sum F_x = R - F = 0$ $R = F$ $\sum M_o = Fh - Q_2L = 0$ $Q_2 = \frac{h}{L}F$	Panel D $\sum M_o = 0$ $X_D b - Fh = 0$ $X_D = \frac{h}{b}F$	Bar S1	Panel A $\sum M_o = X_A a - \left(\frac{h}{L}F\right)l = 0$ $X_A = \left(\frac{lh}{La}\right)F$	Bar S3 $\sum F_y = 0$ $\frac{h}{L}F - \frac{h}{b}F + X_{S3} = 0$ $X_{S3} = \left(\frac{h}{b} - \frac{h}{L}\right)F$	Panel B $\sum M_o = X_{S3}b - X_B a = 0$ $\left(\frac{h}{b} - \frac{h}{L}\right)Fb - X_B a = 0$ $X_B = \frac{b}{a}\left(\frac{h}{b} - \frac{h}{L}\right)F$
$R=F, Q_2=0.4F$					

(2) (10점)

$$\sigma_{cr} = k \frac{E\pi^2}{12(1-\nu^2)(b/t)^2} = 935443.3(t/b)^2$$

panel A: $\sigma_{cr,A} = 935443.3(0.89/100)^2 = 74.1MPa \rightarrow P_{cr,A} = \sigma_{cr,A}(bt) = 6595N$

panel B: $\sigma_{cr,B} = 935443.3(0.89/100)^2 = 74.1MPa \rightarrow P_{cr,B} = \sigma_{cr,B}(bt) = 6595N$

panel C: $\sigma_{cr,C} = 935443.3(0.89/100)^2 = 74.1MPa \rightarrow P_{cr,C} = \sigma_{cr,C}(bt) = 6595N$

panel D: $\sigma_{cr,D} = 935443.3(1/100)^2 = 93.5MPa \rightarrow P_{cr,D} = \sigma_{cr,D}(bt) = 9350N$

(3) 동일 하중 F에 대해 판넬 D에 걸리는 하중은 2F, B에 걸리는 하중은 1.6F로 1.25배이다. 그러나 허용하중의 경우 D 판넬의 9350N이 B 판넬의 6595N의 1.42배이므로 최종적으로 B 판넬이 가장 취약하다. (5점)

(4)

$$\frac{1}{2}F\delta = \sum_{\text{all surfaces}} \frac{1}{2}q_i^2 \left[\frac{\text{panel area}}{(Gt)} \right]_i \quad (5 \text{ pts}) \quad q_i = (\text{edge force} / \text{edge length}), \quad G = \frac{E}{2(1+\nu)} = 79.6GPa$$

$$\begin{aligned} \delta &= \frac{1}{F} \left[\left(\frac{Q_2}{a}\right)^2 \frac{al}{Gt_A} + \left(\frac{X_B}{a}\right)^2 \frac{al}{Gt_B} + \left(\frac{Q_2}{a}\right)^2 \frac{a(L-l-b)}{Gt_C} + \left(\frac{F}{b}\right)^2 \frac{bh}{Gt_D} \right] \\ &= \frac{1}{GF} \left[\left(\frac{hF}{aL}\right)^2 \frac{al}{t_A} + \left(\frac{b}{a^2}\left(\frac{h}{b} - \frac{h}{L}\right)F\right)^2 \frac{al}{t_B} + \left(\frac{hF}{aL}\right)^2 \frac{a(L-l-b)}{t_C} + \left(\frac{F}{b}\right)^2 \frac{bh}{t_D} \right] \\ &= \frac{F}{G} \left[\frac{h^2}{aL^2} \frac{l}{t_A} + \frac{b^2}{a^3} \left(\frac{h}{b} - \frac{h}{L}\right)^2 \frac{l}{t_B} + \left(\frac{h^2}{aL^2}\right) \frac{(L-l-b)}{t_C} + \frac{1}{b} \frac{h}{t_D} \right] \quad (5 \text{ pts}) \end{aligned}$$

5.

$$(1) \text{ (5 pts)} \left\{ \begin{aligned} F_w &= \frac{Fw}{2h} = \frac{(8000N)(1500mm)}{2(1300mm)} = 4615.4N \\ F_h &= \frac{F}{2} = 4000N \\ F_L &= \frac{FL}{2h} = \frac{(8000N)(3000mm)}{2(1300mm)} = 9230.8N \end{aligned} \right.$$

$$(2) \text{ (10 pts)} \left\{ \begin{aligned} e &= \int \frac{\tau \gamma}{2} dV = \int \frac{\tau^2}{2G} dV = \frac{(q/t)^2}{2G} (abt) = q^2 \frac{ab}{2(Gt)} \text{ (5 pts)} \\ \frac{1}{2} T \theta &= \sum_{\text{all surfaces}} \frac{1}{2} q^2 \left[\frac{ab}{(Gt)} \right]_i \xrightarrow{q = \frac{F}{2h} = \frac{T/w}{2h} = \frac{T}{2wh}} \frac{1}{2} T \theta = \sum_{\text{all surfaces}} \frac{1}{2} \left(\frac{T}{2wh} \right)^2 \left[\frac{ab}{(Gt)} \right]_i \\ \theta &= T \left(\frac{1}{2wh} \right)^2 \sum_{\text{all surfaces}} \left[\frac{ab}{(Gt)} \right]_i \rightarrow K = \frac{T}{\theta} = (2wh)^2 \frac{1}{\sum_{\text{all surfaces}} \left[\frac{ab}{(Gt)} \right]_i} \end{aligned} \right.$$

$$(3) \text{ (5 pts)} \left\{ \begin{aligned} \text{Front \& Rear: } \frac{ab}{Gt} &= \frac{(1300mm)(1500mm)}{80000N/mm^2} = 24.375 \\ \text{Top \& Bottom: } \frac{ab}{Gt} &= \frac{(1500mm)(3000mm)}{80000N/mm^2} = 56.25 \\ \text{Right \& Left: } \frac{ab}{Gt} &= \frac{(1300mm)(3000mm)}{80000N/mm^2} = 48.75 \end{aligned} \right.$$

$$K = \frac{T}{\theta} = (2wh)^2 \frac{1}{\sum_{\text{all surfaces}} \left[\frac{ab}{(Gt)} \right]_i} = 4(1500mm)^2 (1300mm)^2 \frac{1}{2(24.375 + 56.25 + 48.75)}$$

$$= 5.88E10 [Nmm / rad] = 1.03E6 [Nm / ^\circ]$$

$$(4) \text{ (10 pts)} \left\{ \begin{aligned} \text{Front \& Rear: } \frac{ab}{Gt} &= \frac{(1300mm)(1500mm)}{80000N/mm^2} = 24.375 \\ \text{Top: } \frac{ab}{Gt} &= \frac{(1500mm)(3000mm)}{0.22(80000N/mm^2)} = 255.7 \\ \text{Bottom: } \frac{ab}{Gt} &= \frac{(1500mm)(3000mm)}{80000N/mm^2} = 56.25 \\ \text{Right \& Left: } \frac{ab}{Gt} &= \frac{(1300mm)(3000mm)}{80000N/mm^2} = 48.75 \end{aligned} \right.$$

$$K = \frac{T}{\theta} = (2wh)^2 \frac{1}{\sum_{\text{all surfaces}} \left[\frac{ab}{(Gt)} \right]_i} = 4(1500mm)^2 (1300mm)^2 \frac{1}{2(24.375 + 48.75) + 255.7 + 56.25}$$

$$= 3.32E10 [Nmm / rad] = 5.79E5 [Nm / ^\circ] \text{ (44\% } \downarrow \text{)}$$